# **Fires, Explosions, and Venting in Nuclear Reactors**

**Palacios A.1, \*, Bradley D.<sup>2</sup>** 

*1 Universidad de las Americas, Puebla, Department of Chemical, Food and Environmental Engineering, Puebla, Mexico 2 University of Leeds, School of Mechanical Engineering, Leeds, UK \*Corresponding author's email: adriana.palacios@udlap.mx*

#### **ABSTRACT**

A brief historical review covers salient reactor fires and explosions, principally centred around the use of graphite as a neutron moderator, and the high temperature generation of hydrogen in reactions of steam and zirconium. An alternative to uncontrolled, excessive, build-up of pressure, followed by uncontrolled explosion, is the provision of a buffer vessel, in which there is separation of hydrogen from radioactive products in permeable membrane separators. The hydrogen is then flared. Possible rates of production of hydrogen are compared, along with the rates at which it can be separated and flared in lifted jet flames, which give the highest burn rates. Cross winds can result in a transition to rim attached, downwash and wake-attached flames, all with a significantly reduced burn rate, or complete flame extinction. The performance of lifted jet flames of  $C_3H_8$ ,  $CH_4$  and  $C_2H_4$ , when exposed to increasing air cross winds velocities, are presented. These provide a basis for synthesising the performance of  $H_2$  flames, also in cross flows. The  $H_2$  relationship is rather different from that of the hydrocarbons, on account of the higher chemical reactivity of hydrogen, its small laminar flame thickness, reduced air requirement, higher acoustic velocity, and minimal flame lift-off distance. Destruction of hydrogen lifted jet flames by the cross flow of atmospheric air is significantly less likely than it is for propane jet flames. Flaring with micro-tubes might be advantageous for integrating flaring with membrane hydrogen separation, whilst high mass flow rates can be achieved with large diameter flares in the lifted flame, supersonic regime.

**KEYWORDS:** Hydrogen, jet flames, reactor venting, cross flow.

#### **NOMENCLATURE**

- *B* molar fuel/cross flow air rate ratio
- *C* molar fraction of air in combined molar flows of fuel and air into lift-off volume
- *C*<sup>*c*</sup> critical value of *C* for reduction in  $U_b^*$  by cross flow
- $C_p$  constant pressure specific heat (J/kg·K)
- *CSL* values of *C*, at the equivalence ratio for maximum laminar burning velocity, *S<sup>L</sup>*
- *D* pipe diameter (m)
- *Do* pipe external diameter (m)
- *f* ratio of fuel to air moles in fuel-air mixture for  $S_L$
- *k* thermal conductivity  $(W/m·K)$
- *L* flame lift-off distance (m)
- $u_i$  mean fuel flow velocity at the exit plane of pipe for subsonic flow. For ratios of atmospheric pressure to  $P_i$  equal to, or less than the critical pressure ratio, or choked sonic velocity after isentropic expansion from  $P_i$  (m/s)
- *U\** dimensionless flow number for choked and unchoked flow,  $(u_f S_L) (\delta_k / D)^{0.4} (P_f P_a)$
- $U_{\delta}$ <sup>\*</sup> Value in Eq. (1) with  $\delta = v/S_L$  in expression for  $U^*$

#### **Greek**

 $\delta_{\nu}$ laminar flame thickness, (m)

 $(k/C_p)_{To}/\rho_j S_L$ 

*φSL* equivalence ratio for maximum laminar

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- $\bar{m}_c$  cross flow air mole density, (moles/m<sup>3</sup>)
- $\bar{m}_j$  fuel mole density, (moles/m<sup>3</sup>)
- *Pa* atmospheric pressure (Pa)
- $P_i$ initial stagnation pressure (Pa)
- *Re<sup>c</sup>* air cross flow Reynolds number, *ucDo/*<sup>ν</sup>*air*
- *SL* maximum laminar burning velocity of the fuel-air mixture in ambient atmosphere  $(m/s)$
- *To* temperature at inner layer of laminar flame (K), see [17]
- $u_c$ cross wind velocity (m/s)

burning velocity

- ν kinematic viscosity, under conditions of ambient atmosphere  $(m^2/s)$
- $\rho$  density (kg/m<sup>3</sup>)

### **Subscripts**

- *a* ambient conditions
- *air* air
- *b* value at blow-off
- *c* cross flow air
- $i$  initial stagnation conditions
- *j* jet fuel

## **INTRODUCTION**

Four key reactor combustion incidents are briefly summarised in Table 1. This is followed by more detailed considerations, which identify key problems, particularly the rapid formation rate of undesirable products at high temperature, their inadequate containment, and the feasibility of venting at an adequate rate to the atmosphere.



#### **Table 1. Some key nuclear reactor fires and explosions**

### **GENERAL CONSIDERATIONS**

The 1957 Windscale fire arose from the use of graphite to moderate neutron energies in the course of plutonium production. Associated Calder Hall power reactors used CO2 as a coolant which, at high temperatures could react with the graphite. The Windscale fire was caused by overheating of the coolant air above 380 °C during Wigner energy release. This caused the graphite to burn in the coolant air. During the fire  $CO<sub>2</sub>$ , was used as an attempted coolant, but it increased the combustion rate. At some risk of excessive  $H_2$  generation,  $H_2O$  was successfully employed [1]. Graphite

moderation was employed at Chernobyl and in the subsequent fire, after loss of control of the reactors, the graphite became incandescent, with the formation of CO. This burned, along with the fuel cladding. Additional difficulties are created in reactors by neutron-induced material degradation [2]. A problem with graphite in such reactors as the Advanced Gas Cooled Reactor, is the neutron displacement damage to the graphite structure. It is difficult to replace the damaged graphite, during the reactor lifetime, although this was achieved with the St. Petersburg reactor. Containment prior to the Chernobyl incident was inadequate.

At Three Mile Island, through loss of coolant, about half the reactor core melted. Radio-nuclides remained inside the reactor or dissolved in water. Reaction of steam with the zirconium cladding of the fuel rods generated  $H_2$ . Corrosion rates of zirconium become 10 times greater inside a reactor [2]. There was no major breach of the containment. Power failure prevented venting of the primary containment up the 100 m high stack. Some of the gases seeped into reactor buildings, accidently exploded, and removed parts of the structure. The widespread dispersal of  $H<sub>2</sub>$  would weaken the mixture, reducing its laminar burning velocity and whilst, initially, turbulence would enhance the burn rate, ultimately it could extinguish it. It has been estimated that a rather lean, nearhomogeneous mixture of  $8\%$  H<sub>2</sub>/air burned, creating an overpressure of 190 kPa [3], and reducing the overall damage. Figure 1, from research sponsored by the UK Atomic Energy Authority at the time, shows the development of flame quenching with increasing turbulence in weak mixtures of 6, 8, and 10% hydrogen (stoichiometric  $% = 29.6$ ) [4].



**Fig. 1.** Quenching effect of the rms turbulent velocity on hydrogen-air flame speed, taken from [4].

After this incident, reactor owners were required to strengthen venting systems to prevent leakage of  $H_2$  into secondary containment buildings. Most of the  $H_2$  was generated from Zircalloy cladding reacting with steam. It has been estimated that  $1,200$  kg of  $H<sub>2</sub>$  would be created, were all the cladding to be be oxidised by steam, and that complete combustion of the zirconium in a 1,000  $MW(e)$  reactor would release  $198'10<sup>9</sup>$  Joules [2].

Hydrogen and  $O_2$  can also be formed in light water cooled nuclear reactors by the radiolytic decomposition of water [5]. If significant amounts of  $H_2$  and  $O_2$  were to be created by radiolysis in stoichiometric proportions, this would be very serious because of the very high reactivity of such a

mixture [6]. However, Gordon et al. [7] found this not to be so, with no more than  $0.7\%$  H<sub>2</sub> created by radiolysis, and this was removed by recombination. To avoid explosive recombination with  $O_2$ , many reactors have been retrofitted with passive hydrogen recombiners within the containment. One approach, implemented in a few Boiling Water Reactors, has been to burn the  $H_2$  inside the containment using distributed glow plugs [8]. Other remedial action has involved injection of  $N<sub>2</sub>$ into the reactor. At Fukushima, reactors survived the earthquake, less so the tsunami. There was no reactor cooling an hour after shut down. At both Three Mile Island and Fukushima there was a failure to remove the radioactive decay heat from the fuel [9].

At Fukushima all the fuel in Unit 1 melted, much of it leaking out. Seawater with neutron absorbing boron were used as coolant, but reactors overheated for many days. The reactors were GE/Toshiba/ Hitachi Boiling Water Reactors, operational since 1971-75, with powers ranging from 460 to 1,784 MW(e). Pressure built up in Units 1 to 3, with most of the fuel melting [9]. Venting was designed to be through an external stack, but, in the absence of power, most of it back-flowed into the top floor of the reactor building. Venting began almost 24 hours into the emergency [10]. Containments were vented to atmosphere. Hydrogen leaked into reactor buildings and caused large explosions in Units 1, 3 and 4. Each Unit is estimated to have produced  $800-1,000$  kg of  $H<sub>2</sub>$ . Hydrogen explosions caused tremendous damage. Even when fissioning had ceased, significant heat was generated through radioactive decay.



**Fig. 2.** Containment of Reactor Pressure Vessel, taken from [9].

As a consequence, the three Fukushima reactor cores, see Fig. 2, melted in the first two or three days of the emergency. There were considerable releases of radio nuclides and cooling water, with a total of ten core melts. The rate of formation of  $H_2$  was controlled by the rate of oxidation of the zirconium fuel cladding by steam, at about 1,300 °C [5]. This rate of reaction was far beyond the capability of  $H_2$  recombiners,  $N_2$  inerting, and the time required to ensure the requisite purity of vented gases. This poses the current major challenge.

# **THE CHALLENGE**

In a loss of coolant, or similar crisis, the reactor and its immediate containment are of inadequate volume to contain all the hydrogen that might be produced, as is evident from the relatively small Primary Reactor Containment shown in Fig. 2. Boiling Water Reactors operate at pressures of about 8 MPa, while Pressurised Water Reactors, with a secondary circuit, operate at about 16 MPa. Were venting to be long delayed a worse situation would arise from failure of the reactor/containment. Unless it is well controlled, allowing emergency venting to atmosphere too early will disperse undesirable radio-nuclides and increase the probability of uncontrolled hydrogen explosions. This perspective leads to the necessity of a large buffer vessel into which the primary products are vented. Ultimately, large amounts of  $H_2$  must be vented, preferably free of undesirable radio-active products. This might be achieved by separating and containing such products, while the hydrogen would be contained and flared in a controlled manner.

Hydrogen separation has been proposed, through the use of hydrogen gas permeable membrane separators in a stream rich in  $H_2$ . This could be passed through a charcoal adsorber to adsorb radioactive particles and then flared in a gas burner [11, 12]. Inability to control the build-up of the high temperature reaction products, inadequate venting rates, particularly of hydrogen, and crisis management have been characteristic features of the described malfunctions. An essential requirement is a large buffering volume to contain the products during their initial high rates of formation. It is also desirable to separate and contain the most damaging products, whilst flaring hydrogen as soon as possible, in order to prevent its build up. A safe balance must be sought between rates of  $H_2$  production, separation, and flaring. Although flaring of  $H_2$  is not essential, if the release is large, it is a safeguard against its hazardous accumulation elsewhere. The hydrogen flaring process is now briefly considered, in terms of its feasibility for achieving adequate burn rates, the practicality of flaring, and the ability of flares to withstand cross winds.

# **CONTROL OF HYDROGEN FLARING**

### **Limitations due to blow-off**

Jet flames exhibit a variety of structures, ranging from lifted flames with high burn rates, in which the fuel jet flow is dominant, to rim and turbulent wake flames, the latter stabilised by the wake of a strong air flow across the fuel pipe. Flame blow-off and extinctions in the former case occur at sufficiently high values of the flow number,  $U^*$ . In the latter, they occur when the Reynolds numbers *Re<sub>c</sub>*, based upon the pipe outer diameter and cross flow air velocity becomes sufficiently high [13]. Between these limits are a variety of other structures ranging from rim-stabilised to downwash flames. Because of the importance of their higher burn rate, lifted flames will be considered in detail, together with the effects of an increasing cross flow of air. The parameter  $U_b^*$ was formulated on the bases of both stretched laminar flamelet mathematical modelling [14] and the experimental derivation, correlation, and validation of appropriate dimensionless groups. Data were drawn from a vast experimental data bank [15]. This covered jet velocities, burning velocities, emitting plume heights, flame lift-off distances, and flame heights, involving six different fuels.

Flaring consists of the burning of a jet of excess fuel in the atmosphere. The highest burn rate within the reaction zone is achieved at the leading edge if the lifted flame, with flamelets burning at the maximum laminar burning velocity,  $S_L$ . The lift-off distance,  $L$ , is the distance between the exit plane of the pipe and this leading edge. If the ratio of fuel pipe diameter, *D*, to laminar flame thickness,  $\delta_k$ , [14] is too small, there is difficulty in maintaining combustion, and the flame is soon quenched by excessive air entrainment.

Figure 3 shows experimentally-based correlations of the dimensionless flow number,  $U_b^*$ , at blowoff, for different values of δ*<sup>k</sup>* /*Db* [16]. There is no cross flow and data are shown for four fuels. The flow number on the x axis, is defined as  $U_b^* = (u/S_L)(\delta_b/D_b)^{0.4}(P_f/P_a)$ , where  $u_j$  is the mean fuel exit velocity,  $D_b$  is the pipe diameter for blow-off, and  $(P/P_a)$  is the ratio of upstream stagnation to atmospheric pressure. Uniquely, in  $H_2$  flames, the high diffusivity of H atoms induces significant heat release earlier in the flame [17]. This necessitates a different approach in the use of flame thickness in generalised correlations [18]. This thickness is given by  $\delta_k = (k/C_p)_{T_o} / \rho_j S_L$  [18] for the leading maximum laminar burning velocity. The data in Fig. 3 are experimentally based, from [16], except for  $C_2H_4$  from [19], and are overwhelmingly from the subsonic pre-choked regime.



**Fig. 3.** Sonic and subsonic lifted jet flame blow-off and quench boundaries, for  $C_3H_8$ ,  $CH_4$ ,  $C_2H_4$ , and  $H_2$ <sub>,</sub> Short dashed horizontal lines show critical pressure ratio condition.

Locations at which the critical pressure ratio is attained on each blow-off curve in Fig. 3 were found from the compatibility of  $\delta_k/D_b$  and  $U_b^*$  at this pressure ratio. These are indicated by the short horizontal broken lines, below which flow is choked at blow-off. Below the blow-off curve, *Ub\**, for a given fuel, towards the lower values of  $\delta_k/D_b$ , is the regime of lifted flames, with larger pipe diameters, above which stable lifted flames can be maintained. Above the curve, is the regime of decreasing pipe diameters, below which blow-off occurs.

Hydrogen flaring, with  $S_L = 3.03$  m/s, is analysed, in terms of these generalised characteristics of lifted jet flames. First, in Fig. 3 the use of a micro-tube,  $D = 2.0$  mm,  $\delta_k = 0.03985$  mm [16, 18],  $\delta_k/D_b = 0.02$ , is considered. The initial/atmospheric pressure ratio is  $P_f/P_a = 1.8$ , just within the subsonic regime, before choked flow develops. Because of the high acoustic velocity of  $H_2$ , arising from the low molecular mass, the exit velocity,  $u_j$ , is also high, at 1,159 m/s, and  $U_b^* = 144$ . These conditions give a micro-tube mass flow rate of  $H_2$  of 1.3 kg/hour, indicated by the upper asterisk in Fig. 3.

Now consider blow-off in the choked flow regime, with  $P/P_a$  increased to 10 and  $D = 10$  mm. In

this regime, the reaction rate is enhanced by shocks and supersonic flows, at high  $U_b^*$ . Now  $\delta_k/D_b =$ 0.004, and  $u_j$  is equal to the acoustic velocity of 1,202 m/s, with an associated density of 0.51924 kg/m<sup>3</sup>. These conditions yield  $U_b^* = 436$ , a mass flow rate of 176.5 kg/hour, indicated by the lower asterisk, and a jet flame heat release rate of 6.9 MW. The generalised data in [15] suggest the jet flame height would be 4.8 m.

Hydrogen has a number of characteristics contributing to high jet velocity flames: a high laminar burning velocity, small flame thickness, small air requirement, and a high acoustic velocity. In contrast, its high reactivity makes it more prone to flame flashback from premixed flames.

#### **LIMITATIONS DUE TO AIR CROSS FLOW**

A further desirable characteristic in the flared venting of jet flames is an ability to survive the cross winds that might occur in the atmosphere. Available experimental data, on the effect of cross wind on  $U_b^*$  have been re-expressed, but for pre-choked flow only, in terms of a parameter *C*. This is the mole fraction of cross flow air in the mixture that is created with the jet fuel, within the lift off distance, *L*. This has a volume  $(\pi D^2/4)L$ , with measured steady fuel jet and air cross flows into it. The resulting experimental data for  $U_b^*$ , processed in this way for  $C_3H_8$ , and CH<sub>4</sub>, from [19], are plotted against values of *C* for different *D*/δ*<sup>k</sup>* , in Fig. 4. Stable, smaller diameter, jet flames exist within the peninsula. At the upper, lifted flame, limit, blow-off occurs, and at the lower limit slower burning flames are no longer lifted, but are attached to the burner.



**Fig. 4.** Blow-off limits of stable CH<sub>4</sub>, C<sub>2</sub>H<sub>4</sub> and C<sub>3</sub>H<sub>8</sub>, lifted flames with air cross flows, in terms of  $U_b^*$  versus *C*, for different values of  $D/\delta_k$ . The dashed curve for H<sub>2</sub> is synthesised from the data of the other fuels.

Initially, as the air cross flow velocity,  $u_c$ , and  $C$  increase from zero, the mixture within the lift-off volume becomes more reactive. This causes the blow-off velocity and *Ub\** to increase. Other similar stability peninsulas for these gases, at different values of  $D/\delta_k$ , can be constructed from the data in [19]. The experiments show that eventually increasing the cross flow necessitated significant reductions in the fuel flow rates and the values of  $U_b^*$  that could be sustained, at a critical value of  $C_1 = C_c$ . This occurs before *C* has attained a value,  $C_{SL}$ , =  $(1 + \varphi_{SL})^{-1}$ , at which the associated equivalence ratio in the lift-off volume is that for  $S_L$ , namely,  $\varphi_{SL}$ . Both of these values of *C* are

given in Table 2 for the different fuels. Of course, additional air is still entrained by the jet downstream beyond the lift-off distance. At  $C_c$ , the flow number  $U_b^*$  must decrease in order to sustain the flame. Ultimately there is a transition to the lower limb of the peninsula and  $u_c$  is much reduced. For smaller values of  $D/\delta_k$ , values of  $C_c$ , become less clearly defined. An example of this is indicated by the dotted curve for  $C_2H_4$  in Fig. 4.

For the C<sub>3</sub>H<sub>8</sub> peninsula in Fig. 4, at  $C_c$  the experimental data show quenching to begin at  $u_j =$ 242 m/s, with  $u_c = 5$  m/s, as  $U_b^*$  begins to fall sharply. For the CH<sub>4</sub> peninsula comparable values are  $u_j$  = 192 m/s and  $u_c$  = 3 m/s. These low values of  $u_c$  suggest that quite moderate cross flows can jeopardise the stability of lifted flames.

No cross wind data could be found for  $H<sub>2</sub>$ , but from the experimental cross wind data available for the other three gases, it was possible to estimate the performance of hydrogen lifted jet flames in air cross flow. Another important factor controlling lifted jet flames in the presence of cross flows, is the ratio, *f*, of fuel to air moles at  $\varphi_{SL}$ . Values of *f* also are given in Table 2. Hydrogen requires significantly less air than hydrocarbons, and the value of  $f$  for  $H_2$  is more than ten times higher than that for  $C_3H_8$ . As can be seen from Table 2, its  $C_{SL}$  value of 0.569, is significantly lower than those for the hydrocarbons. It was found that, for the same value of  $D/\delta_k$ ,  $C_c$  tends to decrease with  $C_{SL}$ . Consequently, the value of  $C_c$  also will be low. Guidance about the extent of this deficit below  $C_{SL}$ for  $H<sub>2</sub>$  was obtained from consideration of that occurring for other gases. This led to the tentative assignment of a value of  $C_c = 0.44$  for H<sub>2</sub>, with a value of  $U_b^*$  of 144 at  $C = 0$ , and  $D/\delta_k = 50$ , taken from Fig. 3. These considerations enabled the tentative, dashed, characteristic peninsula for  $H_2$ , shown in Fig. 4, to be constructed.

From the material balance of jet and cross flows in the lift-off volume, it follows that that *C* =  $1/(B + 1)$ . Here  $B = (u_j \overline{m}_j/u_c \overline{m}_c) \pi/4(D/L)$ , in which the first bracketed term is the ratio of fuel to air molar fluxes into the lift-off volume. Evaluation of  $u/u_c$  requires the normalised lift-off distance for H2 lifted flames, and this was found from the expression given for *L/D*, based on a different expression for flame thickness in  $U_{\delta}$ <sup>\*</sup> [24]:

$$
(L/D)f = -0.0002U_{\delta}^{*2} + 0.19U_{\delta}^{*} - 3.3. \tag{1}
$$

This remains valid as the cross flow develops, but it becomes increasingly unreliable beyond *C<sup>c</sup>* . The expression for *B* yields values of  $u/u_c$ , at the onset of the rapid decline in  $U_b^*$ . The value of  $U_b^*$ then gives that of  $u_j$ .

Fuel	$D/\delta_{\iota}$	Ref. $S_L$	$\varphi_{SL}$	$S_I$ , m/s	$\tau$	$C_{SL}$	$C_c$
$C_3H_8$	60	[20]	1.1	0.43	0.046	0.956	0.93
$C_2H_4$	32	[21]	1.2	0.72	0.084	0.923	0.65
CH <sub>4</sub>	62	[22]	1.02	0.39	0.107	0.903	0.81
H <sub>2</sub>	50	[23]	1.8	3.03	0.756	0.569	(0.44)

**Table 2. Property values and references for characterising Fig. 4** 

These various considerations suggest that for H<sub>2</sub>, with  $U_b^* = 144$  and  $u_j = 1159$  m/s, with  $C_c = 0.44$ ,  $D/\delta_k$  = 50, and  $f = 0.756$ , then  $u_c = 62$  m/s. A natural atmospheric cross wind as high as this is uncommon. It is therefore unlikely that a lifted, venting hydrogen flame could be significantly disturbed by atmospheric conditions, and make the transition to a slower burning attached flame. This behaviour contrasts with that for  $C_3H_8$  and  $CH_4$ . The high value of *f* for hydrogen reduces the air requirement and lift-off distance, while the high acoustic velocity is associated with high jet velocities at a given Mach number. These can be subsonic and in excess of 1,000 m/s.

## **CONCLUSIONS**

1. Loss of coolant and other malfunctions can result in reactors over-heating and creating a variety of chemical reactions and heat releases. This must also be viewed in the context of improving operational efficiencies by operating reactors at higher temperatures. The least acceptable consequence of these is uncontrolled reactor failures, with the release of radioactive products and explosive gases into the atmosphere. Ideally, such a release could be avoided by early venting of the reactor, without any release of noxious products and flammable gases, and no external explosion. This might be achieved by venting the reactor into a much larger buffer vessel, in which the hydrogen might be wholly or partially separated and then flared.

2. In normal operation,  $H_2$  recombiners can process about 195 kg/h of  $H_2$ , but in the case of an accident, the required rate would increase 100 to 400 fold [11], beyond the capabilities of this technique. In this situation, it has been proposed that, after removal of the water from the gaseous mixture, the  $H_2$  should be separated, using a gas permeable membrane separator [11]. The  $H_2$  stream would then pass through a charcoal adsorber to remove radioactive products, before being finally flared. If choked flow flaring on a 10 mm pipe, as demonstrated in Fig. 3 at a rate of 176.5 kg/h, were to be employed, the accumulated approximate estimate of  $3,000 \text{ kg}$  of H<sub>2</sub> at Fukushima would be flared on three such burners in just under 6 hours. There are many ways in which  $H_2$  membrane separation can be implemented, covering a rich variety of materials, and structures [25].

3. With hydrogen permeable membrane separation, it is suggested in [11] that the differential pressure across the membranes should not exceed 1.724 MPa. If the  $H_2$  were to be stored at 2 MPa and 300 K, 1,000 kg of  $H_2$  would occupy a volume of 1,638 m<sup>3</sup>, a cube with a 11.8 m side. This is a practically convenient size, which might combine storage and separation. The present analyses of the subsequent flaring have shown that the characteristics of  $H_2$  are particularly well suited to a flexible approach to storage, separation and flaring, albeit with some possible delay for  $H_2$ separation if dispersal of harmful radio-active products is to be avoided.

4. Flaring of  $H_2$  is favoured by its low air requirements which leads to compact lifted flames. Its high acoustic velocities, arising from its low molecular mass, combined with its high burning velocity, lead to high values of fuel jet velocity. Although the analysis of air cross flow on  $H_2$  lifted flames provides only an estimate of velocities, rather than accurate predictions, it nevertheless clearly shows that the extinction of lifted flames due to atmospheric cross winds is unlikely. The same cannot be said of  $C_3H_8$  and  $CH_4$  flames. There is clearly a need for experimental data on  $H_2$ lifted jet flames in cross flows.

5. A unique aspect of  $H_2$  jet flames is their ability to support micro-jet flames, a consequence of their low  $\delta_k$  values. This could be relevant also in the separation process. Another possibility is for vented gas from the reactor to be immediately flared in micro-jets, followed by removal of radionuclides.

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### **REFERENCES**

- [1] The Windscale Inquiry, The National Archives, 1978.
- [2] S.J. Zinkle, G.S. Was, Materials Challenges in Nuclear Energy, Acta Mater. 61 (2013) 735–758.
- [3] H.A. Postma, Analysis of Three Mile Island (TMI-2) Hydrogen Burn, Thermal Hydraulics of Nuclear Reactors, 1983.
- [4] D. Bradley, K.J. Al-Khishali, S.F. Hall, Turbulent Combustion in Near-Limit Hydrogen-Air Mixtures, Combust. Flame 54 (1983) 61–70.
- [5] Safety of Nuclear Reactors, World Nuclear Association, Updated May 2016.
- [6] D. Bradley, M. Shehata, Acceleration of Laminar Hydrogen/Oxygen Flames in a Tube and the Possible Onset of Detonation, Int. J. Hydrogen Energ. 43 (2018) 6734–6744.
- [7] S. Gordon, K.H. Schmidt, J.R. Honekamp, An Analysis of the Hydrogen Bubble Concerns in the Three Mile Island Unit-2 Reactor Vessel, Radiat. Phys. Chem. 21 (1983) 247–258.
- [8] F. Tamanini, E.A.Ural, J.L. Chaffee, Hydrogen Combustion Experiments in a 1/4 Scale Model of a Nuclear Power Plant Containment, Proc. Combust. Inst. 22 (1988) 1715–1722.
- [9] Fukushima Accident, World Nuclear Association, Updated October 2017.
- [10] M. Holt, R.J. Campbell, M.B. Nikitin, Fukushima Nuclear Disaster, Congressional Research Service Report for Congress., 7–5700, 2012.
- [11] V.M. Callaghan, E.P. Flynn, B.M. Pokora, Containment Hydrogen Removal System for a Nuclear Power Plant, United States Patent 4430293 (1984).
- [12] Lei Wang, Cheng Shao, Hai Wang, Hong Wu, Radial Basis Function Neural Networks-Based Modeling of the Membrane Separation Process: Hydrogen Recovery from Refinery Gases, Jour. Nat. Gas Chem. 15 (2006) 230–234.
- [13] M.R. Johnson, L.W, Kostiuk, Efficiencies of Low-Momentum Jet Diffusion Flames in Crosswinds, Combust. Flame 123 (2000) 189–200.
- [14] D. Bradley, P.H. Gaskell, Xiao-Jun Gu, The Mathematical Modeling of Liftoff and Blowoff of Turbulent Non-Premixed Methane Jet Flames at High Strain Rates, Proc. Combust. Inst. 27 (1998) 1199–1206.
- [15] D. Bradley, P.H. Gaskell, X.-J. Gu, A. Palacios, Jet Flame Heights, Lift-Off Distances and Mean Flame Surface Density for Extensive Ranges of Fuels and Flow Rates, Combust. Flame 164 (2016) 400–409.
- [16] A. Palacios, D. Bradley, Generalised Correlations of Blow-Off and Flame Quenching for Sub-Sonic and Choked Jet Flames, Combust. Flame 185 (2017) 309–318.
- [17] D. Bradley, S.E-D. Habik, S.A. El-Sherif, A Generalisation of Laminar Burning Velocities and Volumetric Heat Release Rates, Combust. Flame 87 (1991) 336–345.
- [18] J. Göttgens, F. Mauss, N. Peters, Analytic Approximations of Burning Velocities and Flame Thicknesses of Lean Hydrogen, Methane, Ethylene, Ethane, Acetylene and Propane Flames, Proc. Combust. Inst. 24 (1992) 129–135.
- [19] G.T. Kalghatgi, Blow-Out Stability of Gaseous Jet Diffusion Flames Part I: In Still Air, Combust. Sci. Tech. 26 (1981) 233–239.
- [20] A. Vanmaaren, L.P.H. DeGoey, Stretch and the Adiabatic Burning Velocity of Methane- and Propane-Air Flames, Combust. Sci. and Tech. 102 (1994) 309–314.
- [21] T. Hirasawa, C.J. Sung, A. Joshi, Z. Yang, H. Wang, C.K. Law, Determination of Laminar Flame Speeds using Digital Particle Image Velocimetry: Binary Fuel Blends of Ethylene, Proc. Combust. Inst. 29 (2002) 1427–1434.
- [22] Xiao-Jun Gu, M.Z. Haq, M. Lawes, R. Wooley, Laminar Burning Velocity and Markstein Lengths of Methane–Air Mixtures, Combust. Flame 121 (2000) 41–58.
- [23] C.J. Sun, C.J. Sung, L. He, C.K. Law, Dynamics of Weakly Stretched Flames: Quantitative Description and Extraction of Global Flame Parameters, Combust. Flame 118 (1999) 108–128.
- [24] A. Palacios, J. Casal, D. Bradley, Prediction of Lift-Off Distance in Choked and Subsonic Hydrogen Jet Fires, Catal. Today (2017) https:doi.org10.1016j.cattod.2017.11.024.
- [25] N.W. Ockwig, T.M. Nenoff, Membranes for Hydrogen Separation, Chem. Rev. 107 (2007) 4078–4110.